

Transport and Chemical Effects on Concurrent and Opposed-Flow Flame Spread at Microgravity

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INTRODUCTION

Flame spread over flat solid fuel beds is a useful means of understanding more complex two-phase non-premixed spreading flames, such as those that may occur due to accidents in inhabited buildings and orbiting spacecraft. The role of buoyant convection on flame spread is substantial, especially for thermally-thick fuels. With suitable assumptions, deRis [1] showed that the spread rate (S_f) is proportional to the buoyant or forced convection velocity (U) and thus suggests that S_f is indeterminate at μg (since $S_f = U$) unless a forced flow is applied. (In contrast, for thermally thin fuels, the ideal S_f is independent of U [1].) The conventional view [2], as supported by computations and space experiments, is that for quiescent μg conditions, S_f must be unsteady and decreasing until extinction occurs due to radiative losses. However, this view does not consider that radiative transfer to the fuel surface can enhance flame spread. In recent work [3] we have found evidence that radiative transfer from the flame itself can lead to steady flame spread at μg over thick fuel beds. Our current work focuses on refining these experiments and a companion modeling effort toward the goal of a space flight experiment called Radiative Enhancement Effects on Flame Spread (REEFS) planned for the International Space Station (ISS) c. 2007.

OPPOSED FLOW EXPERIMENTS AND MODELING

Microgravity experiments on flame spread over thermally-thick fuels were conducted using foam fuels to obtain low density and thermal conductivity, and thus large spread rate (S_f) over thermally-thick fuels compared to dense fuels such as PMMA. This scheme enabled meaningful results to be obtained even in 2.2 second drop tower experiments. After evaluating numerous candidate materials, we chose open-cell polyphenolic foams primarily because they have lower sooting tendency and negligible melting or dripping tendency compared to other foams such as polystyrene or polyurethane. Experiments were conducted in an apparatus consisting of a 20 liter combustion chamber that was filled with the desired atmosphere by a computer-controlled partial pressure gas mixing system. The samples are ignited by an electrically heated wire than in turn ignited a sheet of nitrocellulose. The flames were imaged using CCD cameras whose signals were connected via fiber-optic cables to ground-based S-VHS video recorders. Both direct video and interferometry were employed. The igniter was controlled and the radiometer data were collected by a microcomputer data acquisition and control system.

Figure 1 shows that for CO_2 -diluted atmospheres the steady values of S_f could be higher at μg than $1g$, especially at low O_2 concentrations, but for N_2 -diluted atmospheres S_f was always higher at $1g$ than μg . At μg , S_f can actually be higher in CO_2 than N_2 at the same O_2 concentration even though CO_2 has a larger C_p and thus yields lower T_f than N_2 for the same O_2 concentration. Figure 1 also shows that for CO_2 diluent the minimum O_2 concentration supporting combustion is substantially lower at μg than $1g$, whereas for N_2 the minimum O_2 concentration is higher at μg . All of these results show that flames in CO_2 -diluted atmospheres burn more robustly at μg than $1g$ whereas the opposite trend is found for N_2 . This is likely due to three factors. First, $\bar{\sigma}$ is larger for O_2 - CO_2 atmospheres (since both the combustion products and ambient atmosphere contain radiant species, whereas for O_2 - N_2 only the combustion products radiate) which increases the heat flux to the fuel bed and thus S_f . Second, without buoyant convection ($U = 0$), the flame thickness $\bar{\delta}_g = \bar{\sigma}_g / (U + S_f)$ is much thicker at μg than at $1g$, thus μg flames have more volume and can transfer more radiation to fuel bed. Interferometer images (not shown) confirm that flames are much thicker at μg . This effect is more important for lower O_2 concentrations (thus lower S_f) which explains why the difference between $1g$ and μg spread rates in O_2 - CO_2 atmospheres is larger at lower O_2

concentrations. Third, O₂-CO₂ atmospheres can reabsorb and re-radiate emitted radiation whereas O₂-N₂ atmospheres cannot, thus substantial radiative heat losses that would otherwise occur at μ g with thick flames in strongly radiating O₂-CO₂ atmospheres are at least partially suppressed.

Figure 2 shows predicted pressure effects on S_f [3] for O₂-CO₂ atmospheres at μ g. Two different assumed values of T_f are shown. All gas properties are evaluated at the average temperature $(T_f+T_\infty)/2$. The model provides reasonable S_f estimates except near the low-pressure extinction limit, where heat losses may dominate, leading to slower than predicted flame spread, and at high pressure where there may be a transition to nearly opaque conditions. The opacity at high pressure causes less radiative transfer thus less fuel bed preheating than predicted by our optically-thin model, which leads to lower S_f than the optically-thin predictions. (Neither of these effects can be predicted by the simple optically-thin, loss-free model developed in [3].)

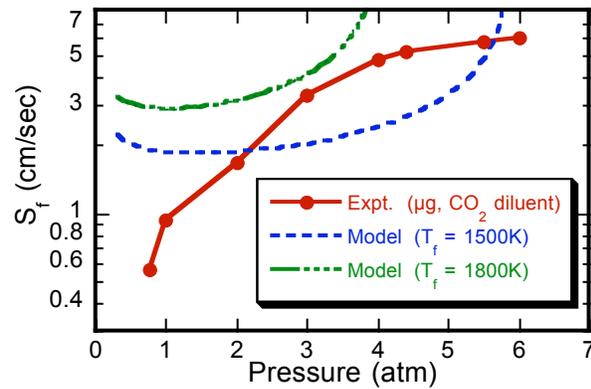
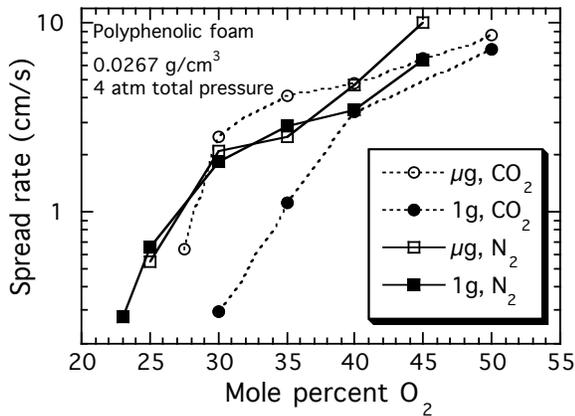


Figure 1. Effect of oxygen concentration on spread rates over thick solid fuel beds at microgravity and earth gravity, polyphenolic foam, density: 0.0267 g/cm³, 4 atm pressure

Figure 2. Effect of pressure on spread rate over thick solid fuel beds at microgravity for polyphenolic foam, density: 0.0267 g/cm³, 40%O₂-60%CO₂.

Figures 3a and 3b show the radiative characteristics of flame spread in O₂-CO₂ mixtures at 1g and μ g, respectively. Narrow-angle thermopile-type radiometers mounted 10 cm from the fuel bed were used to determine radiative emissions from the flames. Three radiometers were used: (1) a front-side (burning side) radiometer viewing a hole in the fuel bed to measure only the outward gas-phase radiative loss, (2) another front-side radiometer viewing the fuel surface to measure both the outward gas-phase and surface radiative fluxes, thus total radiative loss, (3) a back-side radiometer viewing through the hole to measure the inward gas-phase radiative heat flux, thus the fuel bed heating due to gas-phase radiation. The only case where the back-side radiometer (which measures the gas-phase radiant heat flux to the fuel bed) shows comparable intensity and timing with two front radiometers is for the O₂-CO₂ atmosphere at μ g (Fig. 3b). This is likely because only in this case is there substantial emission, absorption and re-emission, which is the only means to obtain substantial radiative flux to the rear-side radiometer. O₂-N₂ atmospheres (not shown) do not show this behavior at all, and even for O₂-CO₂ atmospheres this is seen only at μ g (Fig. 3b) where the flame thickness is larger and thus the total radiative flux is greater. Note also that the gas-phase radiative loss ("Front, gas only" curves) at μ g is actually lower for O₂-CO₂ than O₂-N₂ atmospheres due to reabsorption by the ambient atmosphere for O₂-CO₂. At 1g, (Fig. 3a) the surface radiation is much larger than gas-phase radiation due to the decreased flame thickness thus decreased volume of radiating gas at 1g. These results confirm our hypotheses concerning radiative transfer effects on μ g flame spread, in particular that (1) radiative preheating of the fuel bed by the gas is significant in radiatively-active atmospheres at μ g, (2) reabsorption effects can prevent massive heat losses (thus extinction) in radiatively-active atmospheres at μ g and (3) these effects are less important at 1g due to substantial U caused by buoyancy which leads to smaller flame thicknesses thus less volume of radiating gas.

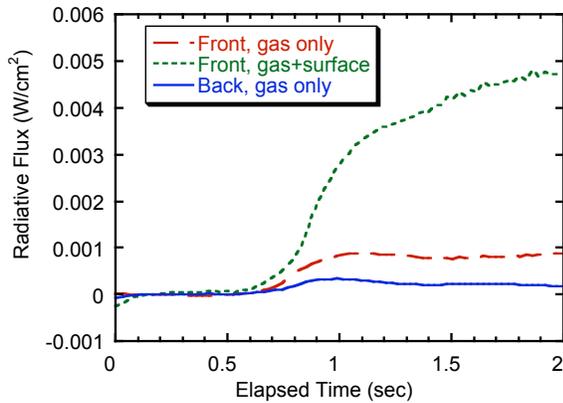


Figure 3. Radiative flux characteristics of flames spreading over polyphenolic foam fuel. (a) 40% O₂ - 60% CO₂, earth gravity.

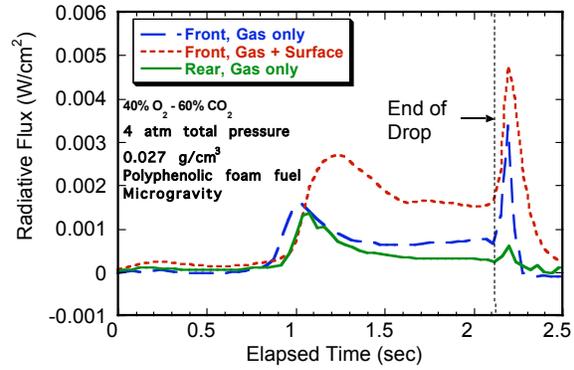


Figure 3. Radiative flux characteristics of flames spreading over polyphenolic foam fuel. (b) 40% O₂ - 60% CO₂, microgravity.

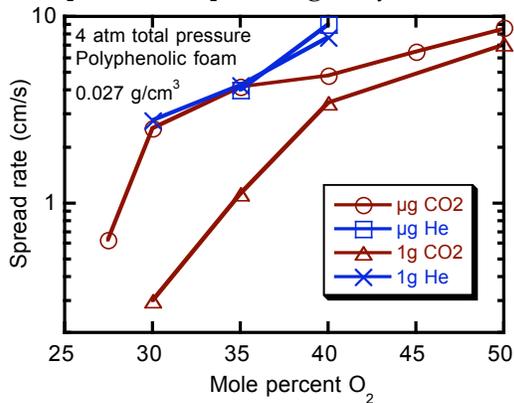


Figure 4. Effect of oxygen concentration on spread rates over thick solid fuel beds at μg and earth gravity – comparison of CO₂ and helium as a fire extinguishing agent.

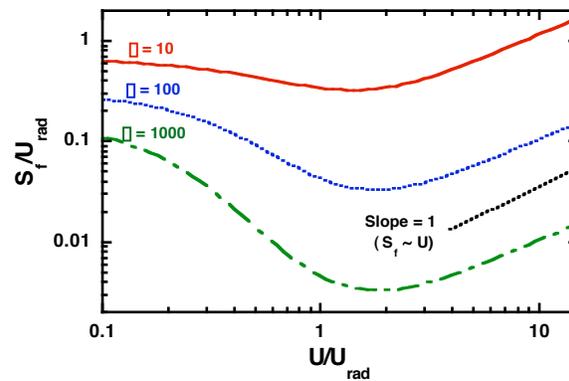


Figure 5. Predicted combined effects of convection and radiation on flame spread rates over thick solid fuel beds

Several tests were conducted using He diluent at 4 atm (Fig. 4). It was found that He- and CO₂-diluted atmospheres exhibit nearly the same S_f , for a given O₂ mole fraction even though CO₂ has a mole-based C_p at 300K that is 1.8 times higher than He, and at 2000K is 3.9 times higher. Also, He has a thermal conductivity (κ_g) 9.4 times higher than CO₂. Both of these factors should lead to higher T_f and S_f in He than in CO₂-diluted atmospheres at the same O₂ concentration. Furthermore, at 1g the minimum O₂ mole fraction supporting combustion was nearly the same (30%) in He and CO₂-diluted atmospheres whereas at μg the minimum O₂ mole fraction was much higher for He (35%) than CO₂ (27%). There are at least three reasons for the observed behavior. First, the Lewis number of O₂ in He is much higher than O₂ in CO₂ (≈ 1.20 vs. ≈ 0.84), which leads to lower spread rates for He [4]. Second, the higher κ_g and ρ_g of He leads to thicker flames and thus greater radiative loss for the same S_f since the heat loss per unit volume depends only on the radiating combustion products whose concentrations are not significantly different (at fixed ambient O₂ concentration) from O₂-CO₂ atmospheres. Third, unlike CO₂, He is radiatively non-participating and thus no reabsorption or re-emission occurs. Consequently, we conclude that He may be a superior inerting agent at μg on several bases. First, at μg He is more effective than CO₂ on a mole basis (thus pressure times storage volume basis), meaning that the size and weight of storage bottles would be smaller for the same fire-fighting capability. Second, He is much more effective on a mass basis (by about 11x) at μg . Third, He has no physiological activity, unlike CO₂ which affects human respiration. These results are relevant to fire safety in manned spacecraft, particularly the International Space Station that uses CO₂ fire extinguishers.

A model was developed that included the combined effects of convection and radiation on flame spread rates over thick solid fuel beds. Figure 5 shows model predictions in terms of the thick-fuel

flame spread parameter β and the opposed forced flow velocity U referenced to a characteristic radiative velocity U_{rad} . It can be seen that there is an intermediate velocity U that minimizes the spread rate. At high U the classical thick-fuel behavior predicted by deRis is observed and at sufficiently low U , the spread rate becomes independent of U (the radiation-dominated regime). Note that in the high- U limit, S_f is proportional to $1/\beta$ whereas at low U , the effect of β is much weaker.

CONCURRENT FLOW RESULTS AT EARTH GRAVITY

Buoyancy-driven upward flame spread over thermally thick fuels is of great practical importance because it is a paradigm for the main mechanism of fire spread in most building fires. Previously a model for upward flame spread over thermally-thin fuels, including the effects of transverse heat and momentum losses to the sides of the fuel samples and surface radiative losses was developed. Such losses were necessary to yield steady flame lengths and spread rates and with such losses, model predictions were found to agree well with experiments [5]. Recently we conducted analogous experiments using low-density but thermally-thick foam fuels, but instead of steady spread, these tests yielded, for reasons still not clear to us, unsteady behavior due to a “jumping” mode of flame spread. These behaviors were found to occur for a wide range of fuel types, pressures, oxygen mole fractions and sample widths and were found not only near extinction limit but also far from the limit conditions. The spatial and temporal intervals of hopping were very regular for moderately narrow samples, whereas wider samples exhibited an unsteady “walking” or “ratcheting” mode and flames ignited on narrower samples extinguished. For thermally-thin fuels similar behavior was found but only near extinction limits. Figure 6 shows a sequence of images illustrating the “jumping” behavior for a narrow sample. Fig. 7 shows the corresponding temporal history of the leading and trailing edges of the flame. It can be seen that the leading edge progresses at a relatively constant rate as if a “source” of flame length, with the trailing edge pausing then “jumping” to catch up. Current work is focused on determine the mechanism(s) responsible for this unusual behavior.



Figure 6. “Jumping” Flame spread, at 45%O₂-CO₂, 1atm, width=10mm, thickness=12mm

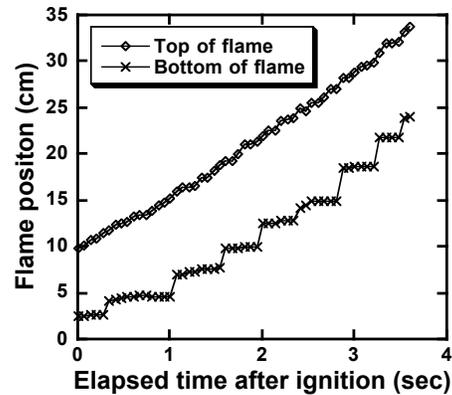


Figure 7. Flame Top and Bottom position as Elapsed time at 45%O₂-CO₂, 1atm, width=10mm, thickness=12mm

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